1. Carroll & Ostlie, Problem 2.8:

- (a) The Hubble Space Telescope is in a nearly circular orbit, approximately 610 km above the surface of Earth. Estimate its orbital period. (2 pts)
- (b) Communications and weather satellites are often placed in *geosynchronous* "parking" orbits above Earth. These are orbits where satellites can remain fixed above a specific point on the surface of Earth. At what altitude must these satellites be located? (3 pts)
- (c) Is it possible for a satellite in a geosynchronous orbit to remain "parked" over any location on the surface of Earth? Why or why not? (2 pts)
- 2. The energy of a planet traveling at velocity v around the Sun can be expressed as

$$\frac{1}{2}mv^2 - \frac{GM_{\odot}m}{r} = Cm,$$

where C is a constant.

- (a) Show that $C = -GM_{\odot}/2a$ for an arbitrary orbit, where a is the semi-major axis of the orbital path. (5 pts) *Hint: Since the total energy is constant along the entire orbit, evaluate it at a convenient point.*
- (b) Use your result from (a) to obtain an expression for the speed of a planet at an arbitrary point along its orbit. What are the maximum and minimum speeds for the Earth, for Mars, and for Jupiter? (5 pts)
- 3. You are orbiting the Earth in your spacecraft at an altitude of 100 km, and receive clearance from Houston to fire the spacecraft engines and produce a tangential velocity which will take you to the Moon. (Assume that the spacecraft's orbit is always in the Moon's orbital plane.)
 - (a) In a two-body approximation (Earth, spacecraft) how much extra velocity (in km sec⁻¹) must be added to put the spacecraft onto an orbit whose apocenter is at the Moon? (5 pts)
 - (b) How much extra velocity would you need to escape from the Earth altogether? (5 pts)